AN ADAPTIVE UNSTRUCTURED TRI-TREE ITERATIVE SOLVER FOR MIXED FINITE ELEMENT FORMULATION **OF** THE **STOKES EQUATIONS**

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SUMMARY

An iterative adaptive equation solver for solving the implicit Stokes equations simultaneously with hi-tree grid generation is developed. The **tri-tree** grid generator builds a hierarchical grid **structure** which is mapped to a finite element grid at each hierarchical level. For each hierarchical finite element grid the Stokes equations are solved. The approximate solution *at* each level is projected onto the next finer grid and **used as** a *start* **vector** for the iterative equation solver at the finer level. When the finest grid is reached, the equation solver is iterated until a tolerated solution is reached.

In order to reduce the overall work, the element matrices are integrated analytically beforehand. The efficiency and behaviour of the present adaptive method **are** compared with those of the previously developed iterative equation solver which is preconditioned by incomplete **LU** factorization with coupled node fill-in.

The efficiency of the incomplete coupled node fill-in preconditioner is shown to **be** largely dependent on the global node numbering. The preconditioner is therefore tested for the natural node **ordering** of the **tri-tree** grid generator and for different ways of sorting the nodes.

KEY WORDS: grid generation; **tri-tree; unstructured grid; finite elements; mixed formulation; analytic integration; adaptive solver; Stokes equations**

INTRODUCTION

Intensive research on developing efficient algorithms for solving the Navier- Stokes equations for arbitrary geometries **has** taken place in several physical disciplines such **as** aerodynamics,' hydrodynamics² and haemodynamics.³ For implicit solution algorithms,⁴⁻⁶ direct equation solvers have shown limitations due to rather large computer storage and computer time requirements.⁷ In view of **this,** iterative equation solvers have been paid extensive attention, with the ultimate goal to be able to solve the Navier-Stokes equations for large, time-dependent, three-dimensional problems with complex geometry. **Although** there have **been** substantial developments towards efficient solvers, there **are** still needs and possibilities for further improvements.

Recently, several iterative equation solvers for non-symmetric equation systems have been developed and tested.⁸⁻¹² These iterative equation solvers have gained quite a lot in both efficiency and robustness by the use of different preconditioning algorithms of the equation system.¹³⁻¹⁵ In previous papers^{5,8} a new incomplete LU factorization preconditioner with a coupled node fill-in algorithm **was** presented. The philosophy of **this ILU** preconditioner made it possible to obtain also a preconditioning **matrix** for the pressure coefficients in the equation **matrix.** Fill-ins with **this** algorithm were allowed where the nodes in the equation system were coupled and not only where the

CCC **0271-2091/96/090899-15** *0* ¹⁹⁹⁶**by** John Wiley & Sons, Ltd. *Received October 1994 Revised August I995* coefficients were initially different fiom zero. This ILU preconditioner revealed advantageous properties also when the equation system was reduced to form an inner-outer iterative algorithm.⁸

In the present work the global order of node numbering was found to play an important role in the convergence rate. Several node-ordering sequences have therefore been tested. **An** advantageous nodeordering scheme seems to be to number the nodes in such a way that during the incompleted elimination there will be no more contribution **to** the element **matrix** from already eliminated nodes. The most efficient node ordering for incomplete coupled node fill-in preconditioning seems to be to number the nodes from the periphery of the finite element grid towards the centre of the grid in an increasing number sequence.

The most time-consuming operation in iterative equation solvers of the conjugate gradient type is matrix-vector multiplication. Since the finite element equations are solved approximately for successively finer grids during the refinement procedure, the matrix generation of the equation system should be **as** fast **as** possible. Traditionally, numerical integration is applied to form the equation **matrix.** However, since simple elements such **as** triangles in two dimensions and tetrahedra in three dimensions **are** applied, the integration of the element matrix terms can be executed analytically. Analytical integration will then save a lot of computational work during the finite element calculations. The integration formulae consist of a constant part, independent of element size, multiplied by a term containing the relative location of the nodes within each element.

During the transition from coarse to finer grid the solution of the coarse grid is interpolated to the fine grid and used **as** a start vector at the fine grid. The refinement procedure on the grid consists of dividing each element into four new elements in two dimensions and eight new elements in three dimensions. Then some nodes will be common to both the coarse and the fine grid. For these nodes, solution values of the coarse grid **are used** directly. New nodes in the fine grid **are** generated at the midpoints between the nodes in the coarse grid. The start values for the iterations at these points in the fine grid **are** then found by linear interpolation. The main purpose of the present adaptive algorithm is to obtain better start vectors **as** the **grids** become more and more refined. When the finest grid is reached, the solution is iterated until the desired convergence criterion is satisfied.

In a previous paper a new tri-tree method¹⁶ for generating unstructured grids, $17,18$ the tri-tree algorithm for generating **grids** in two and three dimensions, was presented. The tri-tree algorithm method starts with a triangle or tetrahedron which is subdivided into four new triangles or eight new tetrahedra respectively. The tri-tree structure then has pointers like the quad-tree and oct-tree.^{19,20} The **main** and essential difference is that the leaves in the tri-tree consist of triangles and tetrahedra. The triangulation procedure of the tri-tree element structure is then much simplified compared with that of the oct-tree structure and will only consist of connecting triangles or tetrahedra of different sizes. **By** introducing **very** mild restrictions on the tri-tree structure, which hardly affect the ability of local refinements, the triangulation procedure becomes very simple. The elements generated **are** optimal in the sense that they do not collapse during the refinements. The elements **are** equilateral triangles and tetrahedra, or at the interfaces of elements of different sizes the equilateral triangles will be divided into two and the equilateral tetrahedra will **be** divided **into** two or four.

During the triangulation procedure an efficient search algorithm is needed for finding co-ordinate **points** in **space.** In the present work a lexical tree search algorithm for the point co-ordinates **has** proved **to** be very efficient.

The initial triangle is successively subdivided into four new triangles and the tetrahedron into eight new tetrahedra. The successive subdivision is continued until the required level of refinement is reached. At each level of tri-tree refinement an associated finite element grid can be constructed and used for finite element calculations. The **tri-tree** data structure is therefore well suited for an adaptive algorithm.

The approximate solution procedure for the Navier-Stokes equations is more complicated **than** for other positive definite **systems** owing **to** the **zero** diagonal block in the equation matrix. In previous papers^{5,8} the coupled node fill-in LU factorization was designed and applied as preconditioning for the Stokes and Navier-Stokes equations. In the same papers the Bi-CGSTAB conjugate gradient algorithm proved to work well also for equation systems which were not positive definite. In the Bi-CGSTAB smoothing algorithm the element matrices **are** only needed in matrix-vector multiplication. The **matxix**vector product can be done node by node for **each** element and the equation matrix need not be assembled and stored. The matrix coefficients are generated whenever needed.

EQUATIONS

The Stokes equations are linear and are given by

$$
-\mu \nabla^2 \mathbf{v} + \nabla p = 0 \quad \text{in} \quad \Omega,
$$
 (1)

$$
-\nabla \cdot \mathbf{v} = 0 \quad \text{in} \quad \Omega, \tag{2}
$$

where **v** is the velocity vector, p is the pressure and μ is the viscosity coefficient. The first equation is the equation of motion which contains a diffusion and a pressure gradient term. The second equation is the equation of continuity. A minus sign is introduced in the continuity equation in order to obtain the same sign for the pressure gradient **as** for the continuity equation in the finite element formulation. In the finite element formulation, two different orders of basis functions are applied for approximating velocities and pressure. With the first-order basis functions the velocities are approximated by linear polynomials and the pressure is considered constant on each element. With the second-order basis functions the velocites **are** approximated with quadratic basis functions and the pressure is approximated with linear basis functions on each element.²¹ The Babuska-Brezzi condition is satisfied for both these finite element formulations. Denote the quadratic polynomials by N_i , the linear polynomials by L_i and the constant polynomial on each element by K_c. Then by the Galerkin residual method and integration by parts the first-order finite element formulation of the Stokes equation system becomes

$$
\mathbf{F}_{\mathbf{v}} = \int_{\Omega} \mu \nabla L_i \cdot \nabla \mathbf{v} \, d\Omega - \int_{\Omega} \nabla L_i p \, d\Omega - \int_{\delta \Omega} \mu L_i \frac{\partial \mathbf{v}}{\partial n} \, d\delta \Omega + \int_{\delta \Omega} L_i p \, d\delta \Omega = 0,
$$
\n
$$
\mathbf{F}_{\mathbf{p}} = - \int_{\Omega} K_{\mathbf{c}} \nabla \cdot \mathbf{v} \, d\Omega = 0.
$$
\n(3)

The second-order finite element formulation of the Stokes equation system becomes

$$
\mathbf{F}_{\mathbf{v}} = \int_{\Omega} \mu \nabla N_i \cdot \nabla \mathbf{v} \, d\Omega - \int_{\Omega} \nabla N_i p \, d\Omega - \int_{\delta \Omega} \mu N_i \frac{\partial \mathbf{v}}{\partial n} \, d\delta \Omega + \int_{\delta \Omega} N_i p \, d\delta \Omega = 0,
$$
\n
$$
\mathbf{F}_{\mathbf{p}} = - \int_{\Omega} L_i \nabla \cdot \mathbf{v} \, d\Omega = 0.
$$
\n(4)

The following equation system *can* then be solved for the first-order formulation:

$$
\left[\begin{aligned}\n\int_{\Omega} \mu \nabla L_i \nabla L_j \, d\Omega & -\int_{\Omega} \nabla L_i K_{\epsilon} \, d\Omega \\
-\int_{\Omega} K_{\epsilon} \nabla L_j \, d\Omega & 0\n\end{aligned}\right]\n\left[\begin{aligned}\n\mathbf{v} \\
\mathbf{p}\n\end{aligned}\right] = -\n\left[\begin{aligned}\n\int_{\partial \Omega} \left(-\mu L_i \frac{\partial \mathbf{v}}{\partial n} + L_i p\right) d\delta \Omega \\
0\n\end{aligned}\right].\n\tag{5}
$$

For the second-order formulation the equation system becomes

$$
\left[\begin{bmatrix} \int_{\Omega} \mu \nabla N_i \nabla N_j d\Omega & -\int_{\Omega} \nabla N_i L_j d\Omega \\ -\int_{\Omega} L_i \nabla N_j d\Omega & 0 \end{bmatrix} \begin{bmatrix} \mathbf{v} \\ \mathbf{p} \end{bmatrix} = -\left[\begin{bmatrix} \int_{\partial \Omega} \left(-\mu N_i \frac{\partial \mathbf{v}}{\partial n} + N_i p\right) d\delta \Omega \\ 0 \end{bmatrix} \right].
$$
 (6)

ANALYTIC INTEGRATION

Let the linear basis functions be denoted by L_i and the quadratic basis functions by N_i . Then in three dimensions

$$
L_i = a_i + b_i x + c_i y + d_i z.
$$

The quadratic basis function can then be given **as** a function of the linear basis function. For the comer nodes *i* and midside nodes *n* respectively

$$
N_i = L_i(2L_i - 1), \qquad N_n = 4L_jL_k,
$$

where the nodes *j* and *k are* the comer nodes on each side of the midside node *n.* The local numbering of the nodes is shown in Figure *2* (see next section). The comer nodes **are** numbered first, then the midside nodes.

Let n_d be the spatial dimension. The exact integrals can be computed by the formula

$$
\int_A L_i^{\alpha} L_j^{\beta} L_k^{\gamma} L_l^{\delta} = \frac{\alpha! \beta! \gamma!}{(\alpha + \beta + \gamma + \delta + n_d)!} n_d! \Omega.
$$

The **integrals** appearing in the matrices then become

$$
\int_{\Omega} L_i d\Omega = \frac{\Omega}{n_d + 1},
$$

$$
\int_{\Omega} L_i^2 d\Omega = \frac{2\Omega}{(n_d + 1)(n_d + 2)},
$$

$$
\int_{\Omega} L_i L_j d\Omega = \frac{\Omega}{(n_d + 1)(n_d + 2)}, \quad i \neq j.
$$

In the formulae below the δ -function is defined by

$$
\delta_{ij} = \begin{cases} 1, & i = j, \\ 0, & i \neq j. \end{cases}
$$

the formulae below the δ -function is defined by
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Let the corner nodes have the local node numbers $1, \ldots, n_c$ and let the midside nodes be locally numbered as $n_c + 1, \ldots, n_c$. In the first-order basis function formulation the integrals of the derivatives in the equation **matrix** *are* given by

 \mathbf{r}

 \mathbf{r}

$$
i \leq n_c, \qquad j \leq n_c,
$$

$$
\int_{\Omega} \frac{\partial L_i}{\partial x} \frac{\partial L_j}{\partial x} d\Omega = \frac{\partial L_i}{\partial x} \frac{\partial L_j}{\partial x} \Omega,
$$

$$
\int_{\Omega} K_e \frac{\partial L_j}{\partial x} d\Omega = K_e \frac{\partial L_j}{\partial x} \Omega.
$$
 (7)

In the second-order basis function formulation the integrals of the derivatives in the equation matrix **are given** by

$$
i \le n_c, \qquad j \le n_c,
$$
\n
$$
\int_{\Omega} \frac{\partial N_i}{\partial x} \frac{\partial N_j}{\partial x} d\Omega = [16(\delta_{ij} + 1) - 8(n_d + 2) + (n_d + 1)(n_d + 2)] \frac{\partial L_i}{\partial x} \frac{\partial L_j}{\partial x} \frac{\partial L_j}{\partial x} \frac{\partial L_j}{\partial x} d\Omega
$$
\n
$$
\int_{\Omega} L_i \frac{\partial N_j}{\partial x} d\Omega = [4(\delta_{ij} + 1) - (n_d + 2)] \frac{\partial L_j}{\partial x} \frac{\Omega}{(n_d + 1)(n_d + 2)},
$$
\n
$$
n \le n_c, \qquad n_c < m \le n_c,
$$
\n
$$
\frac{\partial N_n}{\partial x} \frac{\partial N_m}{\partial x} d\Omega = 4 \frac{\partial L_n}{\partial x} \left[4(\delta_{nj} + 1) \frac{\partial L_j}{\partial x} + 4(\delta_{ni} + 1) \frac{\partial L_j}{\partial x} - \left(\frac{\partial L_i}{\partial x} + \frac{\partial L_j}{\partial x} \right) (n_d + 2) \right] \frac{\Omega}{(n_d + 1)(n_d + 2)},
$$
\n
$$
\int_{\Omega} L_n \frac{\partial N_m}{\partial x} d\Omega = 4 \left((\delta_{ni} + 1) \frac{\partial L_j}{\partial x} + (\delta_{nj} + 1) \frac{\partial L_j}{\partial x} \right) \frac{\Omega}{(n_d + 1)(n_d + 2)},
$$
\n
$$
n_c < p \le n_c, \qquad n_c < q \le n_c,
$$
\n
$$
\int_{\Omega} \frac{\partial N_p}{\partial x} \frac{\partial N_q}{\partial x} d\Omega = 16 \left((\delta_{in} + 1) \frac{\partial L_j}{\partial x} \frac{\partial L_m}{\partial x} + (\delta_{jn} + 1) \frac{\partial L_i}{\partial x} \frac{\partial L_m}{\partial x} \right) \frac{\Omega}{(n_d + 1)(n_d + 2)}.
$$
\n
$$
+ (\delta_{im} + 1) \frac{\partial L_j}{\partial x} \frac{\partial L_n}{\partial x} + (\delta_{jm} + 1) \frac{\partial L_i}{\partial x} \frac{\partial L_n}{\partial x} \right) \frac{\Omega}{(n_d + 1)(n_d + 2)}
$$

Usually, numerical integration, e.g. **Gauss** integration, is applied to compute the coefficients in the finite element matrices. When simple elements such **as** triangles and tetrahedra **are used,** it is possible to perform analytical integration. The coefficients of diffusion, continuity and pressure gradient *can* be computed exactly. For second-order polynomial approximation, all these texms **are** integrals of secondorder polynomials.

TRI-TREE STRUCTURE

In the tri-tree search algorithm.¹⁶ equilateral triangles and tetrahedra are used as basic domains. The equilateral triangles and tetrahedra **are** then subdivided into new equilateral triangles and tetrahedra. In two dimensions an equilateral triangle is divided into four triangles. A diagram of the two-dimensional tri-tree structure is shown in Figure **1.** *An* initial equilateral triangle is divided into four **new** equilateral triangles. Each of these triangles can then be divided into another four equilateral triangles, and **so on.** The tree structure of these divisions is shown in the lower part of Figure **1.** The record belonging to each triangle contains pointers to the triangles into which it is subdivided. **This** triangulation procedure therefore **permits** local refinements required by the geometric **shape** of the **boundary as well as** the properties of the solution.

In three dimensions an equilateral tetrahedron is divided into eight tetrahedra. The ordering of successive divisions is **organized as** a **tree** *structure.* The **tree record** structure **needs** nine integers in **two** dimensions and **14** integers in three dimensions in **order** to **keep** the **necessary information** *at* each level of subdivision.

The **records** describing each two-dimensional triangular leaf **are** shown in Figure **2. A** level **number** indicates the **size** of division and all triangles or tetrahedra of equal *size* will have the same level **number.**

Figwe 1. Hierarchical **stnrcaue** of **the** tri-lree. *An* **initial equilateral triangle is** divided **into four** new equilateral **triangles.** Each of these **triangles** *can* **thm be** divided **into another four equilateral triangles. and** *80* **OIL The** slructm of **these** divisions is shown in the lower part of the figure. The record belonging to each triangle contains pointers to the triangles into which it is subdivided. **This triangulation procedure** therefore **permits** local **refinements** requid **by the geometric shape of the** boundary **as well as the propexties** of the **solution**

When a division is terminal, the level number is given a negative sign. In addition to the level number, a point index **to** each of the comers of the structure is stored. **This** is not strictly necessary, because the *co***ordinates** of each point *can* be calculated when they **are** needed. However, if the comer points **are** stored, the computing time is considerably reduced. The next positions in the structured record **are** pointers to the records of the divisions. When a triangle or tetrahedron is terminal, some of these pointers **are** used **as** pointers to the neighbouring triangles and tetrahedra instead. The last integer in the record points to the record of the parent triangle or tetrahedron. It is therefore possible to perform both up and **down** searches **in** the tri-tree.

When a triangle or tetrahedron is divided, the midpoint on each line between the comers is calculated. **This** point may already exist if the neighbour has a larger level number. If a point does not exist, it is added to the list of points. In order to be able to search for and add points fast, the list is organized **as** a binary **tree.** The binary tree, Figure 3, is sorted lexically on the point co-ordinates.

In order to find the neighbours of a tri-tree element, a search in the tri-tree is performed to find which hi-tree element contains a point slightly outside the edge or side of the present triangle or tetmhedron. The point to use in the tri-tree search is given by

$$
P = P_{\mathbf{g}} + (P_{\mathbf{g}} - P_{\mathbf{c}})/d + \varepsilon (P_{\mathbf{g}} - P_{\mathbf{c}}). \tag{11}
$$

Figure 2. Numbering of triangular leaves in the tree structure together with global numbering of nodes. The record of each triangular structure contains information on the level of refinement at which the triangle is located. If the refinement level number is negative, the triangle is terminal in the tree structure. The following three numbers in the record point to the co-ordinates of the corners of the triangle. For a non-terminal triangular leaf the next four numbers point to the record of the four triangles into which it is divided. If the triangular leaf is terminal, three of these numbers are used as pointers to the records of neighbouring triangles. The last number in the triangle record points to the record of its parent

In this expression, P_g is the centre of gravity and P_g is a corner in the tri-tree element. The spatial dimension is d ($d=2$ or 3) and ε is a small constant which depends on the accuracy of the actual computer. If ε is zero, P is the point where the line from the corner P_c through the point of gravity hits the opposite edge or side. For small ε the point P will be on the line from corner through the point of gravity slightly outside the tri-tree element. The constant ε should be chosen so that the computer representation of

$$
P_{\rm g} + (P_{\rm g} - P_{\rm c})/d \neq P_{\rm g} + (P_{\rm g} - P_{\rm c})/d + \varepsilon (P_{\rm g} - P_{\rm c})
$$
\n(12)

in only two or three of the least significant digits. The point P defined in this way is a point slightly outside the element edge or side opposite to the comer *P,.* A search in the tri-tree for a **tri-tree** element which encloses **a** point *can* either **start** at the mot of the **tree** or at the location of the last search. If the points which **are** searched for **are** introduced in a random fashion, it will be most efficient to **start at** the root of the tree. When a search for the point P defined above is performed, the a priori knowledge is that the point is enclosed in an adjacent tri-tree element. The probability is therefore high that the adjacent **tri-tree** element belongs **to** the same subtree. If the hi-tree element belongs to the same subtree, it **is** faster to **start** the search at the present location, or even **better** at one level above the present location,

Binary tree, lexically **sorted**

Figure 3. **During** the refinement **process** the nodes with co-ordinates **are stored** in **a binary tree.** The key to each node is the *co* **ordinates,** which detmnine whether one node is smaller **or** larger **than** another. The nodes *me* then lexically **sorted** and **a** fast search algorithm will decide whether a point generated during the refinement procedure is already present in the tree structure

than from the root of the **tree.** *On* the average, experiments indicate that it is most efficient to **start** the search at one level above the present. At each level the four triangles in two dimensions and the eight tetrahedra in three dimensions **are** explored to find which one contains the point.

In the balancing procedure a tree element is refined if more than one neighbour is at a smaller level. The balancing procedure is **an** intemtive procedure. After the balancing procedure the tri-tree is valid for triangulation. In two dimensions there is at most one node at the midpoint of one of the edges of the triangles. **This** tri-tree triangle is divided into two finite element triangles. In three dimensions the situation is more complex. Each equilateral tri-tree tetrahedron can either have one node on one of the edges or three nodes at the edges of one of the sides. If there is one node at one edge, the tetrahedron is divided into two. If there are three nodes at the edges of one side, the tetrahedron is divided into four finite element tetrahedra. The triangulation procedure is only applied to tri-tree elements which are inside the computational domain. When the tri-tree is triangulated, the finite elements are kept in a finite element structure and the tri-tree structure is stored to be used later when the grid is **further** adapted to the solution.

ADAPTIVE SOLVER

Let G^k denote the set of grids $\{G^k : k = 1, ..., N\}$, where the grids G^k are in increasingly finer order. Let $x^k \in X^k$ be the set of functions which we require to solve the set of differential equations on the grid G^k . Let the transfer operator from coarse to fine grid be P^k : $x^{k-1} \rightarrow x^k$, where P^k is the prolongation from coarse to fine. Let the set of differential equations to be solved on G^k be given by

$$
\mathbf{F}^k(\mathbf{x}^k) = \mathbf{b}^k. \tag{13}
$$

Let *Smooth*(\bf{x} , $\bf{\hat{x}}$) be a smoothing or approximate solution algorithm defined on every grid G^k , $\bf{\hat{x}}$ the start vector and **x** the smoothed vector. The adaptive algorithm is then defined **by**

Choose
$$
\tilde{\mathbf{x}}^k
$$

\nfor $\{k = 1; k (= N - 1; k + +\}$
\n{
\n $\mathbf{x}^k = \mathbf{x}^k + \mathbf{P}^k(\mathbf{x}^{k-1} - \tilde{\mathbf{x}}^{k-1});$
\n*Smooth*($\mathbf{x}^k, \tilde{\mathbf{x}}^k$);
\n} Solve $\mathbf{F}^N(\mathbf{x}^N) = \mathbf{b}^N$ iteratively.

The initial triangle or tetrahedron is successively refined until the desired refinement level is reached. At each **tri-tree** level of refinement a finite element grid is constructed and the set of differential equations is solved approximately for **this** grid. The approximate solution on one finite element grid level is then interpolated and projected onto the finer grid and used **as** a **start** vector for **this** grid.

The prolongation P^k is the mapping from coarse to fine grid. The values of the common nodes are taken from the coarse grid **and** the values of the new nodes at the midpoints of each side **are** interpolated linearly. The linear interpolation procedure is simply to take the average between **two** comer nodes. The prolongation algorithm is applied in both two and three dimensions. There exist more complicated local smoothing algorithms which take into account several neighbouring nodes. However, **as** local smoothing is followed by global smoothing, **a** simple first-order local smoothing algorithm is sufficient.

The critical part of the adaptive algorithm is the global smoothing method. The special problem which arises with the Navier-Stokes equations is the zero diagonal block^{5,8} associated with the continuity equation, which implies non-positive definiteness of the equation **matrix.** Thus smoothing algorithms such **as** Gauss-Seidel and traditional **ILU** factorization cannot be applied directly **as** smoothing procedure. However, if some rather arbitrary postconditioning²² matrix is used, this limitation can be overcome. The difficulty with non-positive definiteness *can* also be avoided with inner- outer iterations. As the equation matrix is non-symmetric, the usual conjugate gradient type of smoothing cannot be applied either. The introduction of inner-outer iterations and a postconditioning matrix certainly represents **an** increase in superfluous work. In the present work the CGSTAB conjugate gradient method^{11,12} with coupled node fill-in, which is often considered as an iterative equation solver, is used **as** smoother.

The adaptive multigrid algorithm **starts** with the coarsest grid, computing a smoothed or exact solution for **this** grid. **This** solution and the corresponding residual **are** then prolonged to the finer grid. At the finest grid level the solution is determined fully converged. When the equation system is solved for the finest grid, the adaptive cycle is complete.

The smoothing algorithm within each adaptive iteration can be just a few iterations with the CGSTAB smoother or a fully converged solution found by the CGSTAB equation solver. At each grid level the smoothing procedure can be stopped either after a fixed small number of iterations or by a convergence criterion defined by

$$
\frac{\|\mathbf{r}^k\|}{\|\mathbf{x}^k\|} < \varepsilon_{\mathbf{g}},\tag{14}
$$

where r^k is the residual and x^k is the solution vector at grid level k. The complete adaptive iteration is stopped by the same convergence criterion with $\epsilon = 10^{-4}$.

NODE-ORDERING SCHEME

The node numbering for the tri-tree grid generator is shown in Figure 4a. The nodes in the tri-tree generator are numbered as new nodes are introduced during refinement of the grid. The corners are numbered first, and when the final refinement level is reached, the midside nodes are introduced in element order. The other way of node ordering which has been investigated is based on sorting nodes. In the sorting algorithm the nodes *are* sorted with respect to their distance from the centre of the grid. The node which is furthest away from the centre is given the smallest number. In the first ordering scheme, where all nodes **are** sorted, Figure 4b, the nodes **are** sorted regardless of whether they *are* corner or midside nodes. In the second ordering scheme, Figure **4c,** the comer and midside nodes **are** sorted separately and the comer nodes are numbered before the midside nodes. In Figure **4d** the comer and midside nodes are again sorted separately but the midside nodes are numbered before the comer nodes.

NUMERICAL EXPERIMENTS

The test problem is channel flow with the boundary conditions shown in Figure *5.* The velocities *are* set to **zero** at the walls and a parabolic velocity profile is imposed at the inlet. The Reynolds number for the

Figure 4. Differmt methods for global numbering of nodes: **a, the** numbering **of** nodes **achieved by tri-tree grid generator, b, all the nodes are sorted** with **respect to distance from** *ceatrc* of **grid; c,** comer **and** midside nodes **are sorted separately and** comer nodes **nmbcred first; d,** comer **and** midside **nodes are** *sorted* **separately aad midside nodes numbered first**

channel flow is 1000. The successive **grids** used at each refinement level are shown in Figure 6. Experiments are performed for both first- and second-order basis functions approximating the solution. With first-order basis functions the velocities are approximated **by** linear polynomials and the pressure is approximated by a constant on each element. With second-order basis functions the velocities are approximated by quadratic polynomials and the pressure is approximated by linear polynomials on each element. The pressure and velocities are shown in Figure **7** for the second-order polynomial approximation at grid level 3 and Reynolds number **10o0.** At each level of refinement the finite element equatin system is solved iteratively by Bi-CGSTAB preconditioned by incomplete factorization with coupled node fill-in. The original algorithm is obtained by using the **zero** vector **as start** vector for the iterative solution procedure. The projection algorithm uses the projected solution from the coarser grid **as start** vector. The projection algorithm will therefore have a **start** vector which is much closer **to** the solution vector **than** that of the original algorithm. The convergence criterion at all grid level is set to $\varepsilon = 10^{-4}.$ **EXEREMONG EXEREM USING A CONDENS USI**

The numerical experiments are performed for first-order and second-order basis functions. The effect of sorting the nodes for the first-order basis functions is shown in Table **I.** The results in **this** table indicate that the number of iterations necessary to reach convergence is considerably reduced when the nodes are sorted compared with the case when the nodes appear in tri-tree order. For the most refined grids, levels **4** and *5,* the iterations became stagnant and the iterative equation solved had to be restarted.

Table **I1** shows the effect of sorting the nodes for the second-order basis functions. Three ways of sorting the nodes **are** investigated. The first method consists of sorting **all** the nodes regardless of whether they are corner or midside nodes. Compared with tri-tree node ordering, all sorting methods needed fewer iterations for convergence. The number of iterations to convergence for the three sorting methods did not differ significantly. For the most refined **grid,** level *5,* it was also necessary to **restart** the iterations for the second-order basis functions.

The results of using the projection of the solution from the coarser grid **as start** vector for the firstorder basis functions are shown in Table **111.** All the nodes are sorted in these experiments. The velocity solution from the coarser grid is interpolated linearly. However, **as** the pressure is constant on each element, is was difficult to compute a good projection for the pressure which both increased the convergence rate and converged accurately enough to the right solution. In **fact,** the best starting value for the pressure was the zero vector. The behaviour of the equation solver is very much improved **by** the projection algorithm, especially for the finest grids. For the most refined **grids** it was not possible to obtain a converged solution at all without using the projected solution of the velocities **as start** vector.

Figure 5. Test problem of channel flow with **boundary conditions.** The **velocities are ZQD** *at* **the walls. At the** inlet **a parabolic velocity profile is** introduced

Figure 6. Hierarchy of **@ds** used in **computations. The** initial **grid is** shown **at the top and** has **eight finite elements with a total** of **nine corner nodes. At the next level** of **refinement each** of **these elements is** divided into four **new elements. giving a** *total* **of 32 elements and 25 comer nodes. The** *start* vector for **each fins** grid **is the solution fiom the** *coarser* grid for common **nodes. The start values** for new **nodes** *are* found **by linear interpolation**

The simulation results for the projection algorithm **are** shown in Table **IV** For the original solution algorithm where the zero vector is **used as** start vector, all nodes **are** sorted. For the projection algorithm the three different sorting methods are applied. Again the three sorting methods reveal no sigmficant differences with respect **to** convergence rate. The projection algorithm is considerably faster for medium grid size and converges with **as** little **as** one linear iteration for the finest grids. The reason for convergence within one iteration is of course due to the fact that the **start** vector is within the tolerance of the **final** solution. However, in contrast, the original method does not converge at all. The total amount of work in obtaining the solution by the projection method must for completeness include the work in

Figure 7. Pressure and **velocities** of **channel flow** for **Reynolds number** lo00 *at* grid **level** 3. **The pressure is displaybd as** isobars **and the velocities are shown as velocity vectors**

Table **II.** Number of iterations for different **grids** for second-order basis functions. The first column shows the level of refinement of the grid. The second column shows the number of finite elements in each coordinate direction. The third column shows the number of degrees of freedom for each refinement level. The last columns show the number of iterations *to* convergence for the tri-tree node **ordering,** for all nodes, for the comer and midside nodes sorted separately with the comer **nodes** numbered first and for the comer and midside nodes sorted separately with the midside nodes numbered first, respectively

Level	Grid	Deg. free.	Tri-tree	All sort	C/M sort	M/C sort
	2×2	57				
2	4×4	187				
3	8×8	659	19	12	14	
4	16×16	2467	89	40	36	37
	32×32	9539	417_{50}	206_{50}	22950	205_{50}

Table **III.** Number of iterations and computational work for different grids with and without using the projection of the solution **from** coarser grid **as** *start* vector in the iterations. In both *cases* the iterative equation solver **uses** incomplete factorization with **coupled** node fill-in. In these experiments a **set** of first-order basis fimctions is used, with linear approximation for the velocities **and** constant approximation for the pressure on each element. The first column shows the level of grid refinement. The second **column** shows the number of **degrees** of freedom. The number of **original** iterations is found when the zero vector is **used as** *start* vector. The number of projected iterations is obtained **by** *using* the projection of the solution **hm** the coarser grid **as start** vector. The last **two** columns show work in terms of number of multiplications $\times 10^{-3}$. The initial work is performed during the incomplete coupled node fill-in factorization. The iterative **work** is **performed during** one linear iteration. The subscript numbers in the third and fourth columns indicate the number of iterations between restarts of the iterative equation solver

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Table **IV** Same parameter **values as** in Table **III** for second-order basis functions **used** in the element formulation. The velocities **are** approximated **by** quadratic basis functions and the pressure is approximated **by** linear basis functions on each element. The third column shows the number of iterations needed when all nodes **are sorted** using zero *start* vector. In the next three columns the projected solution from the coarser grid is used **as** *start* vector. Three different sorting methods for the projection algorithm **are** applied, namely all nodes sorted, the corner and midside nodes **sorted** separately with the comer nodes numbered first and the comer and midside nodes **sorted** separately with the midside nodes **numbered** first. Note that the number of iterations to obtain the solution increases until level **4** when using the projection solution **as start** vector. Then the number of iteration **starts** to decrease. This phenomenon is explained in the text. The difference sorting methods do not result in a significant difference in number of iterations to convergence. The two last columns show work in terms of number of multiplications \times 10⁻³. The initial work is performing the incomplete coupled node fill-in decomposition. The iterative work is the number of multiplications $\times 10^{-3}$ performed during one linear iteration. The subscript number in the third column indicates the number of iterations between **restarts** of the itemtive equation solver

obtaining the solution for all coarser grids. Even then, the amount of work to be performed is favourable, **as** the **sum** of work executed for the coarser grids is of the same order of magnitude **as** for the finest grid. All experiments in this investigation have been performed on a standard workstation and the computing time is of the order of minutes.

DISCUSSION

The goal of **this** work has been to develop a solution algorithm for the Navier- -Stokes equations which is robust, fast and sparse. The robustness is attached to the implicit solution techniques for the differential equation system. The speed of the algorithm is tied to the computer time needed. The sparsity is linked to the storage requirements of the algorithm. The adaptive method described in this paper seems to some extent to have these properties.

In the present paper an adaptive method for solving the Navier-Stokes equations is developed. The adaptive algorithm may be considered **as** consisting of the following five essential parts: grid generation, adaptive refinement, matrix integration, intergrid transition and adaptive equation solver.

The grid generation is based on the tri-tree algorithm, which permits the construction of a finite element grid at each tree level. The tri-tree algorithm allows for adapting the grid both to irregular geometry and to the solution of the system of differential equations. The matrix generation is executed by analytic integration and is therefore fast enough for the coefficients in the equation matrix to be easily generated whenever needed in the solution algorithm. The transition **from** coarse **to** fine *grids* is direct and linear interpolation is used for the midside nodes. The iterative solver, Bi- **CGSTAB,** allows for zero diagonal blocks in the equation matrix.

The most important property of the adaptive algorithm is that when the grid is sufficiently refined, the **start** vector is within the solution tolerance and only a few iterations are needed. For more complex boundary conditions and geometries when local spatial refinement is needed, this property can be used **to obtain an accurate solution where large gradients in the solution occur.**

Further work with adaptive tri-tree grid structures for irregular grids will consider non-linear adaptive iterative solvers, linear and non-linear multigrid methods, local grid adaptation to discontinuities in both the solution and the boundary **geometry and hyperbolic upwinding schemes.**

ACKNOWLEDGEMENTS

The author is grateful to Britt von Krogh for corrections of **the manuscript and to Olav Dahl for discussions** of **numerical methods.**

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